



Nonlinear dynamic response and vibration of nanocomposite multilayer organic solar cell



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ABSTRACT

In the recent years, organic solar cell (OSC) has attracted much interest of the research community due to its great promise as renewable sources. This paper presents the first analytical approach to investigate the nonlinear dynamic response and vibration of imperfect rectangular nanocomposite multilayer organic solar cell subjected to mechanical loads using the classical plate theory. Nanocomposite organic solar cell consists of five layers of Al, P3HT:PCBM, PEDOT:PSS, IOT and glass. Motion and compatibility equations are derived using the classical plate theory and taking into account the effects of initial geometrical imperfection and geometrical nonlinearity in Von Karman – Donnell sense. The Galerkin method and fourth – order Runge – Kutta method are used to give explicit expressions of natural frequencies, nonlinear frequency – amplitude relation and nonlinear dynamic responses of nanocomposite organic solar cell. The numerical results show the influences of geometrical parameters, the thickness of layers, imperfections, and mechanical loads on the nonlinear dynamic response and nonlinear vibration of nanocomposite organic solar cell.

1. Introduction

Nanocomposite is a multiphase solid material where one of the phases has one, two or three dimensions of less than 100 nm (nm). Recently, nanocomposites are used in a number of fields and new applications such as automobiles, aerospace, injection molded products, coatings, adhesives, fire-retardants, packaging materials, microelectronic packaging, etc. Nanocomposites are usually added nanoparticles and fibers to enhance mechanical strength, toughness and electrical or thermal conductivity. The mechanical behaviors of nanocomposite with reinforced particles and fibers, such as bending, buckling, vibration, etc., have attracted attention of many researchers around the world. He et al. [1] investigated the mechanical behavior of hierarchical Mg matrix nanocomposite with high volume fraction reinforcement. Al-Lafi et al. [2] studied the mechanical responses of polycarbonate and polycarbonate/multi-walled carbon nanotubes to dynamic loadings at low and high velocities impacts were investigated experimentally using an instrumented falling weight impact tester and a split Hopkinson pressure bar, respectively. Asadi and Wang [3,4] investigated the aeroelastic analysis and dynamic stability analysis of functionally graded carbon nanotube reinforced composite beams and cylindrical

shells subjected to aerodynamic load, axial compression and supersonic airflow. In 2015, Duc et al. [5] presented an investigation on the nonlinear dynamic response and vibration of the imperfect laminated three-phase polymer nanocomposite panel resting on elastic foundations and subjected to hydrodynamic loads. Sourri et al. [6] introduced a theoretical study on the piezoresistive response of carbon nanotubes embedded in polymer nanocomposites in an elastic region; and Wu et al. [7] focused on the dynamic instability of functionally graded multilayer nanocomposite beams reinforced with a low content of graphene nanoplatelets and subjected to a combined action of a periodic axial force and a temperature change. Asadi et al. [8] considered the aero thermoelastic behaviors of supersonic functionally graded carbon nanotube reinforced composite flat panels in thermal environments. Further, Wang et al. [9] reported an improved in situ synthesis of PEDOT:PSS@AgNPs hybrid through simultaneous oxidation-reduction reaction between AgNO₃ and EDOT at room temperature to achieve a hierarchical structure with excellent performance. Using a finite element-based multi-scale modeling approach, Ahmadi et al. [10] analyzed the bending, buckling and free vibration of hybrid polymer matrix composites reinforced by carbon fibers and carbon nanotubes. Mehri et al. [11] scrutinized the aeroelastic responses of functionally

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graded carbon nanotube reinforced composite truncated conical curved panels subjected to aerodynamic load and axial compression. Based on the classical plate theory, Duc and Thu [12] presented an analytical approach to investigate the non-linear dynamic response and vibration of an imperfect three-phase laminated nanocomposite plate and cylindrical panel resting on elastic foundations in thermal environments. Wu et al. [13] dealt with the thermal buckling and postbuckling of functionally graded multilayer nanocomposite plates reinforced with a low content of graphene platelets.

Organic solar cell (OSC) is well-known as a renewable energy device owning the light and flexible properties which could be expected as the next generation of the solar cell technology [14–16]. Multilayer composite structure is the key factor for these photovoltaic devices such as nanocomposite composed of multilayer structure with nanoparticles embedded in polymers (NIP) and polymers deposited on nanoporous thin films (PON) for OSC [17]. Interestingly, several nanostructures such as nanoparticle and nanorods are believed to improve the light harvesting process through the localized surface plasmon resonance (LSPR) effect [18–21]. In this work, we have embedded nanocrystalline such as TiO₂ or ZnO inside PEDOT-PSS as well as P3HT:PCBM [22]. On one hand, the higher quenching of the polymer fluorescence observed in presence of TiO₂ nanoparticles indicates that the transfer of the photogenerated electrons to the TiO₂ is more efficient than rods. On the other hand, the absorption spectrum in the visible light induced by nanorods has been increased about 20% more than nanoparticles. Characterization of the nanocomposite films showed that both the current-voltage characteristics and the photoluminescent properties of the nanocomposite materials were significantly enhanced in comparison with the pure polymers. The role of the buffer layer, i.e. ZnO thin film, in OSC structure will also be discussed in this work. The optimal thickness layer of OSC will be simulated by the diode model fitting to JV characteristics. In order to investigate the operating-temperature range for photovoltaic device parameters, a P3HT + nc-TiO₂ composite photoactive layer was prepared. The photovoltaic conversion efficiency was reached a value of 1.72%. The enhancement in the photoelectrical conversion efficiency of the solar composite-based cells is attributed to the presence of nano-hetero junctions of TiO₂/P3HT and ZnO/P3HT:PCBM. For the temperature range of (30–70) °C, the decrease of the open-circuit potential was compensated by an increase of the fill factor; and the increase in the short-circuit current resulted in an overall increase of the energy conversion efficiency. At elevated temperatures of 60–80 °C the efficiency of the pure P3HT-OSC reached a maximum value of 1.6% and 2.1%, respectively. Over this temperature range the efficiency of P3HT-based cell decreased strongly to zero, while for the composite cells it maintained a value as large as 1.2% at a temperature range of (110–140) °C. The static and dynamic stability of OSC under different loads such as the thermal load and mechanical load have been investigated using the classical shear theory [23]. In general, we have shown that the combination of rods and nanoparticles in the nanocomposite material is able to stabilize the composite structures. Recently, the reinforce of polymers using more complex structure such as auxetic layer and piezoelectric layer, nanoparticles not only increases the strength of the material but also enriches the physical properties of nanocomposite structure. The applications of piezoelectric layers and carbon nanotubes in enhancing nonlinear vibration behaviors and suppressing the postbuckling deflection of functionally graded carbon nanotube reinforced composite annular plates is presented [24,25]. For an illustration purpose, the Young's module, Poisson's ratio, bulk module and other elastic modulus of composite materials will also be discussed [26–28].

This paper presents the first analytical approach to investigate the nonlinear dynamic response and vibration of the imperfect nanocomposite multilayer organic solar cell subjected to mechanical loads. By using the Galerkin method and fourth – order Runge – Kutta method, the influences of geometrical parameters, the thickness of layers, imperfections, mechanical loads on the nonlinear dynamic response and

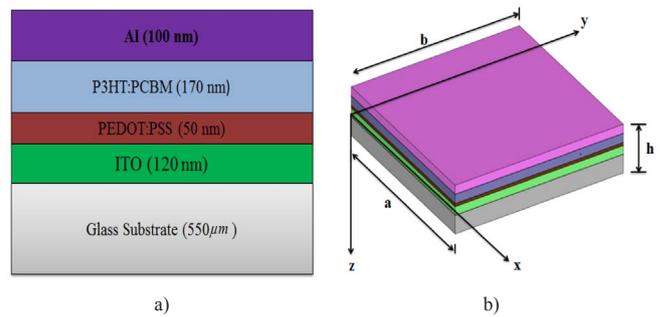


Fig. 1. Geometry and coordinate system of nanocomposite multilayer organic solar cell. (a) 2D model (b) 3D model.

nonlinear vibration of solar cells are discussed in details.

2. The modelling of nanocomposite multilayer organic solar cell

Consider a rectangular nanocomposite organic solar cell with five layers of length of edges a, b and total thickness h as shown in Fig. 1. Five layers are made of Al, P3HT:PCBM, PEDOT:PSS, IOT and glass, respectively. Each layer is assumed to be isotropic and the thickness, Young modulus and Poisson's ratio of each layer are $h_k, E_k,$ and $\nu_k,$ ($k = \overline{1,5}$) respectively. A coordinate system (x,y,z) is established in which (x,y) plane on the middle surface of the cell and z on thickness direction ($-h/2 \leq z \leq h/2$) as shown in Fig. 1.

3. Basic equations

In this study, the classical plate theory is used to derive basic equations to investigate the nonlinear dynamic response and vibration of nanocomposite multilayer organic solar cell.

The strain – displacement relation taking into account the Von Karman non – linear terms are [31–33]

$$\begin{pmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \end{pmatrix} = \begin{pmatrix} \epsilon_x^0 \\ \epsilon_y^0 \\ \gamma_{xy}^0 \end{pmatrix} + z \begin{pmatrix} k_x^1 \\ k_y^1 \\ k_{xy}^1 \end{pmatrix} \quad (1)$$

where

$$\begin{pmatrix} \epsilon_x^0 \\ \epsilon_y^0 \\ \gamma_{xy}^0 \end{pmatrix} = \begin{pmatrix} \frac{\partial u}{\partial x} + \frac{1}{2} \left(\frac{\partial w}{\partial x} \right)^2 \\ \frac{\partial v}{\partial y} + \frac{1}{2} \left(\frac{\partial w}{\partial y} \right)^2 \\ \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} + \frac{\partial w}{\partial x} \frac{\partial w}{\partial y} \end{pmatrix}, \quad \begin{pmatrix} k_x \\ k_y \\ k_{xy} \end{pmatrix} = \begin{pmatrix} -w_{,xx} \\ -w_{,yy} \\ -2w_{,xy} \end{pmatrix}, \quad (2)$$

in which u, v are the displacement components along the x, y directions, respectively.

The stress–strain relationships for the nanocomposite multilayer organic solar cell are defined by the Hooke's law as

$$\begin{pmatrix} \sigma_x \\ \sigma_y \\ \sigma_{xy} \end{pmatrix}_k = \begin{pmatrix} Q'_{11} & Q'_{12} & Q'_{16} \\ Q'_{12} & Q'_{22} & Q'_{26} \\ Q'_{16} & Q'_{26} & Q'_{66} \end{pmatrix}_k \begin{pmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \end{pmatrix}_k, \quad (3)$$

with k is the number of layers of nanocomposite organic solar cell and

$$\begin{aligned} Q'_{11k} &= \frac{E_k}{1-\nu_k^2}, Q'_{12k} = \frac{\nu_k E_k}{1-\nu_k^2}, Q'_{16k} = 0, Q'_{22k} = \frac{E}{1-\nu_k^2}, Q'_{26k} = 0, Q'_{66k} \\ &= \frac{E_k}{2(1 + \nu_k)}. \end{aligned} \quad (4)$$

The force and moment resultants of nanocomposite multilayer organic solar cell are given by

$$N_i = \sum_{k=1}^n \int_{h_{k-1}}^{h_k} [\sigma_i]_k dz, i = x,y,xy, M_i = \sum_{k=1}^n \int_{h_{k-1}}^{h_k} z [\sigma_i]_k dz, i = x,y,xy. \tag{5}$$

Substitution of Eq. (1) into Eq. (3) and the results into Eq. (5) give the constitutive relations as

$$\begin{bmatrix} N_x \\ N_y \\ N_{xy} \\ M_x \\ M_y \\ M_{xy} \end{bmatrix} = \begin{bmatrix} A_{11} & A_{12} & 0 & B_{11} & B_{12} & 0 \\ A_{12} & A_{22} & 0 & B_{12} & B_{22} & 0 \\ 0 & 0 & A_{66} & 0 & 0 & B_{66} \\ B_{11} & B_{12} & 0 & D_{11} & D_{12} & 0 \\ B_{12} & B_{22} & 0 & D_{12} & D_{22} & 0 \\ 0 & 0 & B_{66} & 0 & 0 & D_{66} \end{bmatrix} \begin{bmatrix} \epsilon_x^0 \\ \epsilon_y^0 \\ \gamma_{xy}^0 \\ k_x \\ k_y \\ k_{xy} \end{bmatrix}, \tag{6}$$

where the detail of coefficients $X_{11}, X_{12}, X_{22}, X_{66}$ ($X = A, B, D$) may be found in Appendix A.

The nonlinear equilibrium equations of a nanocomposite multilayer organic solar cell based on the classical plate theory are [31–33]

$$N_{x,x} + N_{y,y} = 0, N_{xy,x} + N_{y,y} = 0, M_{x,xx} + 2M_{xy,xy} + M_{y,yy} + N_x w_{,xx} + 2N_{xy} w_{,xy} + N_y w_{,yy} + q = \rho_1 \frac{\partial^2 w}{\partial t^2}, \tag{7}$$

in which the material damping is not taken into account because of its small influence.

The geometrical compatibility equation for an imperfect nanocomposite multilayer organic solar cell is written as [31–33]

$$\epsilon_{x,yy}^0 + \epsilon_{y,xx}^0 - \gamma_{xy,xy}^0 = w_{,xy}^2 - w_{,xx} w_{,yy} + 2w_{,xy} w_{,xy}^* - w_{,xx} w_{,yy}^* - w_{,yy} w_{,xx}^*. \tag{8}$$

$f(x,y)$ is stress function defined by [5,12]

$$N_x = \frac{\partial^2 f}{\partial y^2}, N_y = \frac{\partial^2 f}{\partial x^2}, N_{xy} = -\frac{\partial^2 f}{\partial x \partial y}. \tag{9}$$

From the constitutive relation (6), one can write

$$\epsilon_x^0 = A_{11}^* N_x + A_{12}^* N_y + A_{13}^* k_x + A_{14}^* k_y, \epsilon_y^0 = A_{22}^* N_y + A_{12}^* N_x + A_{23}^* k_x + A_{24}^* k_y, \gamma_{xy}^0 = A_{31}^* N_{xy} + A_{32}^* k_{xy}, \tag{10}$$

with the detail of coefficients A_{li}^* ($i = \overline{1,4}$), A_{2j}^* ($j = \overline{2,4}$), A_{3k}^* ($k = 1,2$) are given in Appendix A.

By substituting Eq. (10) into Eq. (6) and then into Eq. (7) with the aid of Eq. (8), the system of motion Eq. (7) is rewritten as follows

$$T_{11} f_{,xxxx} + T_{12} f_{,yyyy} + T_{13} f_{,xyxy} + T_{14} w_{,xxxx} + T_{15} w_{,yyyy} + T_{16} w_{,xyxy} + f_{,yy} w_{,xx} - 2f_{,xy} w_{,xy} + f_{,xx} w_{,yy} + q = \rho_1 \frac{\partial^2 w}{\partial t^2}. \tag{11}$$

where the linear operators T_j ($j = \overline{1,6}$) are given Appendix A.

For an imperfect nanocomposite multilayer organic solar cell, Eq. (11) is modified into form as

$$T_{11} f_{,xxxx} + T_{12} f_{,yyyy} + T_{13} f_{,xyxy} + T_{14} (w_{,xxxx} + w_{,xxxx}^*) + T_{15} (w_{,yyyy} + w_{,yyyy}^*) + T_{16} (w_{,xyxy} + w_{,xyxy}^*) + f_{,yy} (w_{,xx} + w_{,xx}^*) - 2f_{,xy} (w_{,xy} + w_{,xy}^*) + f_{,xx} (w_{,yy} + w_{,yy}^*) + q = \rho_1 \frac{\partial^2 w}{\partial t^2}. \tag{12}$$

From Eq. (10) in conjunction with Eq. (9) we have

$$\epsilon_x^0 = A_{11}^* f_{,yy} + A_{12}^* f_{,xx} - A_{13}^* w_{,xx} - A_{14}^* w_{,yy}, \epsilon_y^0 = A_{22}^* f_{,xx} + A_{12}^* f_{,yy} - A_{23}^* w_{,xx} - A_{24}^* w_{,yy}, \gamma_{xy}^0 = -A_{31}^* f_{,xy} - 2A_{32}^* w_{,xy}. \tag{13}$$

Setting Eq. (13) into Eq. (8) gives the compatibility equation of an imperfect nanocomposite multilayer organic solar cell as

$$A_{11}^* f_{,yyyy} + A_{12}^* f_{,xxyy} - A_{13}^* w_{,xxyy} - A_{14}^* w_{,yyyy} + A_{22}^* f_{,xxxx} + A_{12}^* f_{,xxyy} - A_{23}^* w_{,xxxx} - A_{24}^* w_{,xxyy} + A_{31}^* f_{,xxyy} + 2A_{32}^* w_{,xxyy} = w_{,xy}^2 - w_{,xx} w_{,yy} + 2w_{,xy} w_{,xy}^* - w_{,xx} w_{,yy}^* - w_{,yy} w_{,xx}^*. \tag{14}$$

Eqs. (12) and (14) are nonlinear equations in term of variables w and f and used to investigate the nonlinear dynamic response and vibration of imperfect nanocomposite multilayer organic solar cell under mechanical loads using the classical plate theory.

4. Nonlinear vibration analysis

Assume an imperfect nanocomposite multilayer organic solar cell with freely movable edges is subjected to uniformly distributed pressure of intensity q . Thus the boundary conditions are

$$w = N_{xy} = M_x = P_x = 0, N_x = N_{x0} at x = 0, a w = N_{xy} = M_y = P_y = 0, N_y = N_{y0} at y = 0, b \tag{15}$$

where N_{x0}, N_{y0} are in-plane compressive loads at movable edges.

The approximate solutions of w and f satisfying boundary conditions (15) are assumed to be [5,12,29,30,33]

$$w = W \sin \alpha x \sin \beta y, f = A_1 \cos 2\alpha x + A_2 \cos 2\beta y + A_3 \sin \alpha x \sin \beta y + \frac{1}{2} N_{x0} y^2 + \frac{1}{2} N_{y0} x^2, \tag{16}$$

where $\alpha = m\pi/a, \beta = n\pi/b, m, n$ are odd natural numbers representing the number of half waves in the x and y directions, respectively; and $W(t)$ are the time dependent amplitudes.

The initial imperfection w^* is assumed to have the same form of the organic solar cell deflection w , i.e. [5,12,29,30,33]

$$w^*(x,y) = W_0 \sin \alpha x \sin \beta y, \tag{17}$$

in which W_0 is imperfection amplitude of the nanocomposite multilayer organic solar cell.

Introducing Eqs. (16) and (17) into the compatibility equation (14) and solving obtained equation for unknown f leads to

$$A_2 = \frac{(W^2 + 2WW_0)\alpha^2}{32A_{11}^*\beta^2}, A_1 = \frac{(W^2 + 2WW_0)\beta^2}{32A_{22}^*\alpha^2}, A_3 = \frac{A_{14}^*\beta^4 + A_{23}^*\alpha^4 - (2A_{32}^* - A_{13}^* - A_{24}^*)\alpha^2\beta^2}{A_{11}^*\beta^4 + A_{22}^*\alpha^4 + (A_{12}^* + A_{12}^* + A_{31}^*)\alpha^2\beta^2} W. \tag{18}$$

Replacing Eqs. (16)–(18) into Eqs. (12) and then applying Galerkin method to the resulting equations yields

$$S_1(W + 2W_0)W + S_2W + S_3(W^2 + 2WW_0)(W + W_0) - (\alpha^2 N_x^0 + \beta^2 N_y^0)(W + W_0) + S_4(W + W_0)W + S_5(W + W_0) + S_6q = \rho_1 \frac{\partial^2 w}{\partial t^2}. \tag{19}$$

where the detail of coefficients S_i ($i = \overline{1,6}$) are given in Appendix A.

This is basic equation to determine nonlinear dynamic response of a nanocomposite multilayer organic solar cell subjected to mechanical loads.

4.1. Nonlinear dynamic response

Consider a nanocomposite multilayer organic solar cell with freely movable edges only subjected to uniform external pressure $q = Q \sin \Omega t$ (Q is the amplitude of uniformly excited load, Ω is the frequency of the load) and uniform compressive forces P_x and P_y (Pascal) on the edges $x = 0, a$ và $y = 0, b$. In this case, $N_x^0 = -P_x h, N_y^0 = -P_y h$ and Eq. (19) is reduced to

$$S_1(W + 2W_0)W + S_2W + S_3(W^2 + 2WW_0)(W + W_0) + (\alpha^2 P_x - \beta^2 P_y)h(W + W_0) + S_4(W + W_0)W + S_5(W + W_0) + S_6 Q \sin \Omega t = \rho_1 \frac{\partial^2 w}{\partial t^2}. \tag{20}$$

By using Eq. (20), three aspects are taken into consideration: fundamental frequencies of natural vibration of the nanocomposite

multilayer organic solar cell, frequency – amplitude relation of nonlinear free vibration and nonlinear dynamic response of nanocomposite multilayer organic solar cell. The nonlinear dynamic responses of the nanocomposite multilayer organic solar cell can be obtained by solving this equation combined with initial conditions to be assumed as $W(0) = 0, \frac{dW}{dt}(0) = 0$ by using the fourth – order Runge – Kutta method.

4.2. Natural frequency and amplitude – Frequency relation

The fundamental frequencies of a perfect organic solar cell can be determined approximately by an explicit expression from Eq. (20) as

$$\omega_{mn} = \sqrt{-\frac{S_2 + S_5 + (\alpha^2 P_x + \beta^2 P_y)h}{\rho_1}} \tag{21}$$

Consider nonlinear vibration of a perfect organic solar cell, Eq. (20) has of the form

$$\frac{\partial^2 W}{\partial t^2} + \omega_{mn}(W + MW^2 + NW^3) - F \sin(\Omega t) = 0, \tag{22}$$

where

$$M = \frac{S_1 + S_4}{S_2 + S_5 + (\alpha_m^2 P_x + \beta_n^2 P_y)h}, N = \frac{S_3}{S_2 + S_5 + (\alpha_m^2 P_x + \beta_n^2 P_y)h}, F = \frac{S_6 Q}{\rho_1} \tag{23}$$

Seeking solution as $W = A \sin \Omega t$ and applying Galerkin procedure to Eq. (22), the amplitude – frequency relation of nonlinear forced vibration is obtained as

$$\chi^2 - \left(1 - \frac{8}{3\pi} MA + \frac{3}{4} NA^2\right) + \frac{F}{A\omega_{mn}^2} = 0. \tag{24}$$

where

$$\chi = \Omega / \omega_{mn} \tag{25}$$

If $F = 0$, i.e. no excitation acting on the organic solar cell, the frequency –amplitude relation of the free nonlinear vibration is obtained as

$$\omega_{NL}^2 = \omega_{mn}^2 \left(1 - \frac{8}{3\pi} MA + \frac{3}{4} NA^2\right). \tag{26}$$

5. Results and discussion

This section presents the illustrative results for nanocomposite multilayer organic solar cell with the thickness and the material properties of each layer given as in Table 1.

5.1. Natural frequency

The effects of thickness of layers $h_{Al}, h_{PCBM}, h_{PSS}, h_{ITO}$ and modes (m,n) on the natural oscillation frequency of the nanocomposite multilayer organic solar cell are shown in Table 2. It is easy to see that the value of the natural oscillation frequency increases when the thickness of layers increase and the value of modes increase. The Table 2 also

Table 1
The thickness and material properties of the constituent materials of the considered nanocomposite multilayer organic solar cell.

Layers	Thickness	E (GPa)	ν	ρ (kg/m ³)
Al	100 nm	70	0.35	2601
P3HT:PCBM	170 nm	6	0.23	1200
PEDOT:PSS	50 nm	2.3	0.4	1000
ITO	120 nm	116	0.35	7120
Glass Substrate	550 μ m	69	0.23	2400

shows that the effect of ITO layer on natural oscillation frequency of the nanocomposite multilayer organic solar cell is highest and the effect of PSS layer on natural oscillation frequency of the nanocomposite multilayer organic solar cell is lowest in five layers. This conclusion is easy to explain because the elastic modulus of ITO layer is highest and the elastic modulus of PSS layer is lowest of all. The Table 2 also shows that the natural oscillation frequency increases when modes (m,n) increase.

5.2. Dynamic response

5.2.1. Effect of geometrical parameter

Fig. 2 illustrates the effect of geometric factor a/b on nonlinear dynamic response of nanocomposite multilayer organic solar cell with $P_x = 0, P_y = 0$. From Fig. 2, the deflection amplitude of the organic solar cell increases when increasing the ratio a/b . In a fabricating process, the ratio a/b has been chosen as close as a square shape cell. In other words, the best ratio a/b for an experiment is 1.0 [34], hence our finding is consistent with the experiments. This result has reconfirmed the best shape for the organic solar cell sample is a square instead of a rectangular.

5.2.2. Effect of the thickness of layers

Figs. 3–5 indicate the effect of the thickness of PCBM layer, PSS layer and ITO layer on the nonlinear dynamic response of the nanocomposite multilayer organic solar cell with $a = b = 1, P_x = 0, P_y = 0$, respectively. It can be seen that the deflection amplitude of the nonlinear dynamic response of the solar cell decreases when the thickness of these layers increases. Because of the trade-off between the recombination of the electron-hole pairs and the collection of the exciton in organic solar cell, the optimal thickness of the active layer is limited to approximately 200 nm [18].

Hence, our finding is important since the dynamic response of the solar cell does not depend strongly on the thickness of each layer. In other words, we do not need to be worried about stability of the solar cell within a range of the thickness order from 170 to 700 nm for PCBM, from 50 to 900 nm for PSS, from 50 to 135 nm for ITO.

5.2.3. Effect of mechanical loads

Fig. 6 considers the effect of harmonic uniform exciting force with amplitudes $Q = 100 \text{ N/m}^2$ [$Q = 250 \text{ N/m}^2$ and $Q = 300 \text{ N/m}^2$] on the dynamic response of nanocomposite multilayer organic solar cell. From these figures, it is seen that the nonlinear deflection amplitude of the organic solar cell is considerably increased when excitation force amplitude Q increases.

Figs. 7 and 8 show nonlinear dynamic response of the nanocomposite multilayer organic solar cell with various values of the pre-loaded axial compression P_x, P_y . Clearly, the higher value of the pre-loaded axial compression is, the higher nonlinear deflection amplitude of the organic solar cell is.

5.2.4. Effect of imperfection

The effect of initial imperfection W_0 on the dynamic response of nanocomposite multilayer organic solar cell is indicated in Fig. 9. As expected, the reduction of amplitude of initial imperfection makes the deflection amplitude of nonlinear vibration of the organic solar cell decrease.

5.3. Frequency – Amplitude relation

5.3.1. Effect of external force F

Fig. 10 shows the effect of external force F on the frequency – amplitude relations of frequency – amplitude curves of the nanocomposite multilayer organic solar cell. As can be seen, when the excitation force decreases, the curves of forced vibration are closer to the curve of free vibration.

Table 2
Effects of the thickness of layers and modes on natural frequencies of the nanocomposite multilayer organic solar cell.

h_{Al} (nm)	h_{PCBM} (nm)	h_{PSS} (nm)	h_{ITO} (nm)	(m,n)			
				(1,1)	(1,3)	(3,3)	(3,5)
100	170	50	120	19011	71192	171100	286770
120	170	50	120	19733	72940	177600	297540
100	185	50	120	19040	71261	171360	287200
100	170	65	120	19032	71243	171290	287090
100	170	50	135	19808	73121	178270	298660
120	185	65	135	20548	74938	184940	309710
135	195	70	160	22265	79231	200390	335350
150	210	90	170	23227	81679	209050	349720

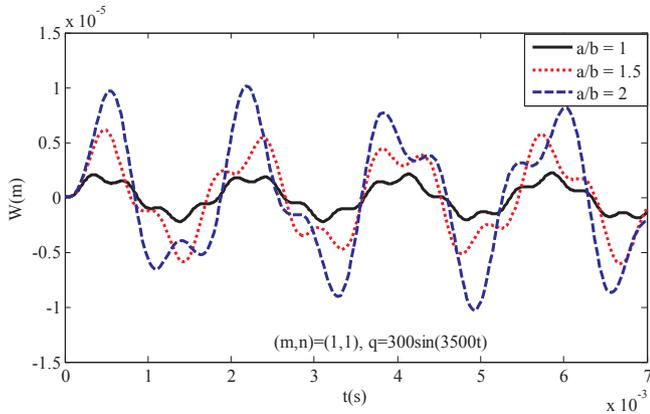


Fig. 2. Effect of ratio a/b on the nonlinear dynamic response of nanocomposite multilayer organic solar cell ($P_x = 0, P_y = 0$).

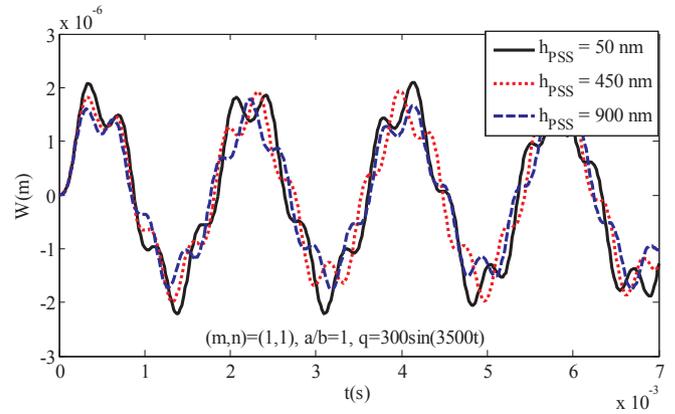


Fig. 4. Effect of the thickness of PSS layer on the dynamic response of nanocomposite multilayer organic solar cell ($P_x = 0, P_y = 0$).

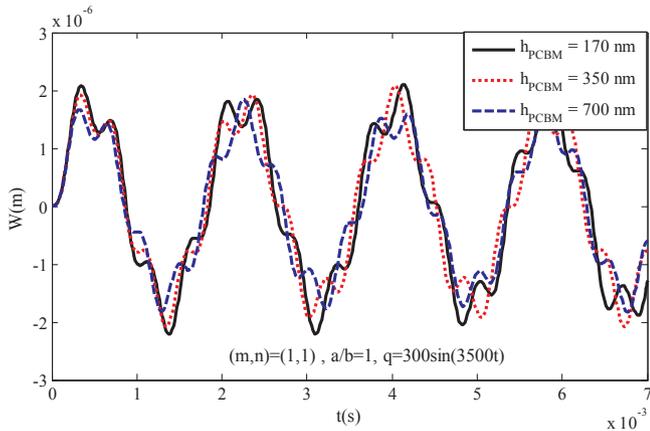


Fig. 3. Effect of the thickness of PCBM layer on the dynamic response of nanocomposite multilayer organic solar cell ($P_x = 0, P_y = 0$).

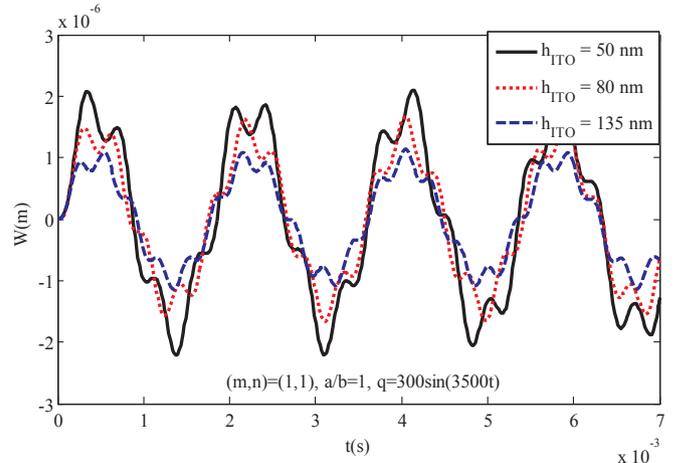


Fig. 5. Effect of the thickness of ITO layer on the dynamic response of nanocomposite multilayer organic solar cell ($P_x = 0, P_y = 0$).

5.3.2. Effect of the thickness of layers

Fig. 11 shows the effects of the thickness of ITO layer on the frequency – amplitude relations of the nanocomposite multilayer organic solar cell. One can see that with the same dynamic amplitude, nanocomposite multilayer organic solar cell with lower thickness have smaller frequency than the nanocomposite multilayer organic solar cell with higher thickness.

6. Conclusions

This paper investigated the nonlinear dynamic response and vibration of the nanocomposite multilayer organic solar cell subjected to mechanical loads based on the classic plate theory taking into account geometrical nonlinearity and initial geometrical imperfection. By using

Galerkin method and fourth – order Runge – Kutta method, the influence of the geometrical parameter, the thickness of layers, initial imperfection and mechanical loads on the nonlinear dynamic response and vibration of the solar cell are examined in detail. Some conclusions can be obtained from the present analysis:

- The thickness of every layer of organic solar cell has a significant influence on the nonlinear dynamic response and nonlinear vibration of the nanocomposite multilayer organic solar cell.
- The effect of ITO layer on nonlinear vibration of the nanocomposite multilayer organic solar cell is highest and the effect of PSS layer on nonlinear vibration of the nanocomposite multilayer organic solar cell is lowest in five layers.

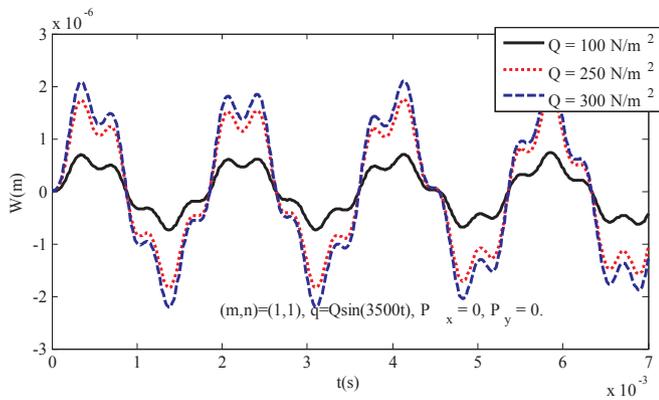


Fig. 6. Effect of the exciting force amplitude Q on the dynamic response of nanocomposite multilayer organic solar cell ($P_x = 0, P_y = 0$).

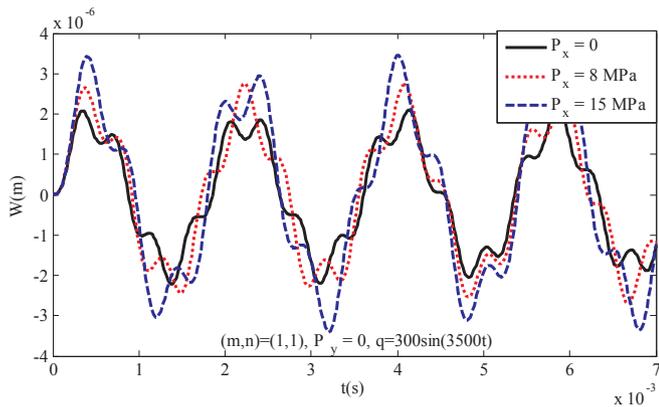


Fig. 7. Effect of the pre-loaded axial compression P_x on the dynamic response of nanocomposite multilayer organic solar cell.

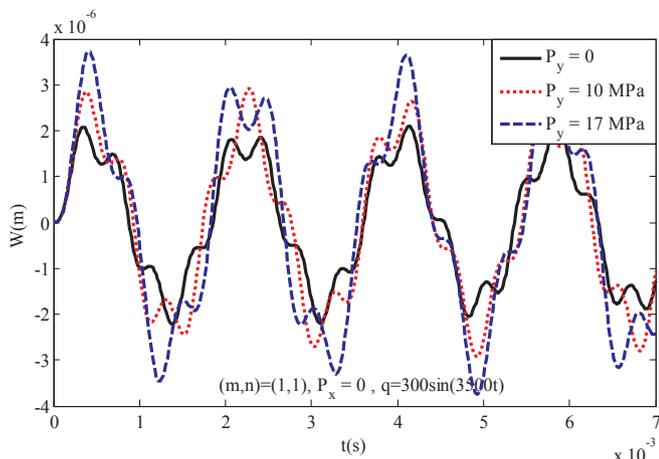


Fig. 8. Effect of the pre-loaded axial compression P_y on the dynamic response of nanocomposite multilayer organic solar cell.

- The nonlinear dynamic amplitude of the nanocomposite solar cell is considerably increased when excitation force amplitude Q and P_x increase.
- Initial geometrical imperfection has negative influence on the nonlinear dynamic response and vibration of nanocomposite multilayer

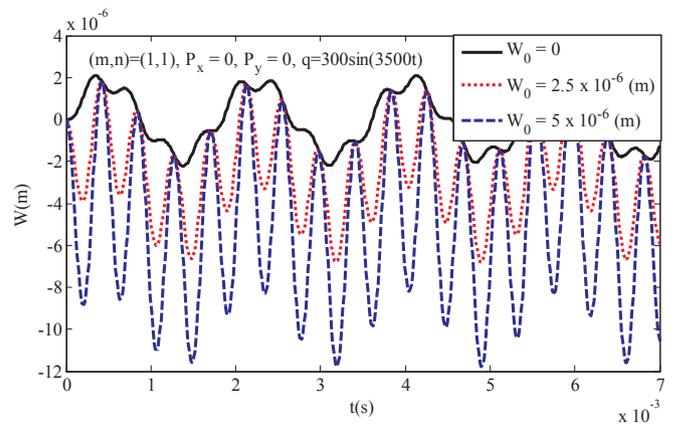


Fig. 9. Effect of initial imperfection W_0 on the dynamic response of nanocomposite multilayer organic solar cell.

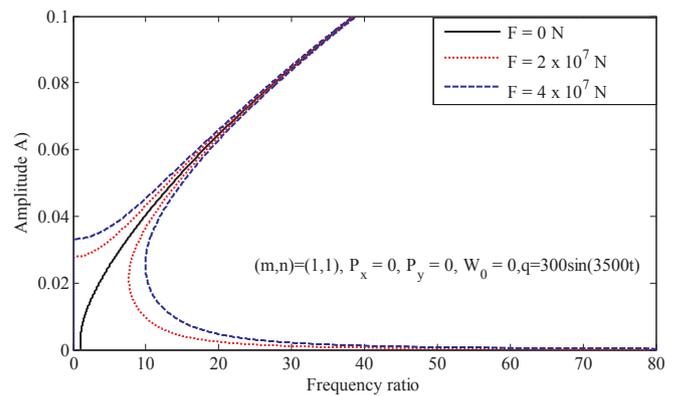


Fig. 10. Effect of external force F on frequency – amplitude curves of the nanocomposite multilayer organic solar cell.

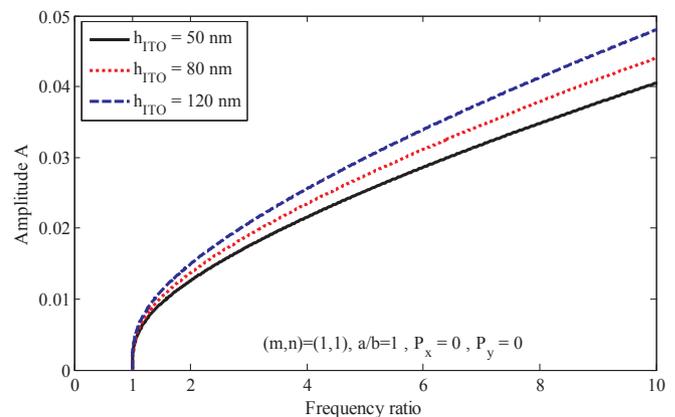


Fig. 11. Effect of the thickness of ITO layer on frequency – amplitude curves of the nanocomposite multilayer organic solar cell.

organic solar cell.

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Appendix A

$$\begin{aligned}
 A_{11} &= \sum_{k=1}^5 \frac{E_k h_k}{1-\nu_k^2}, A_{12} = \sum_{k=1}^5 \frac{\nu_k E_k h_k}{1-\nu_k^2}, A_{22} = A_{11}, \\
 A_{16} &= A_{26} = 0, A_{66} = \sum_{k=1}^5 \frac{E_k h_k}{2(1+\nu_k)}, \\
 B_{11} &= \sum_{k=1}^5 \frac{E_k h_k z_k}{1-\nu_k^2}, B_{12} = \sum_{k=1}^5 \frac{\nu_k E_k h_k z_k}{1-\nu_k^2}, B_{22} = B_{11}, \\
 B_{16} &= B_{26} = 0, B_{66} = \sum_{k=1}^5 \frac{E_k h_k z_k}{2(1+\nu_k)}, \\
 D_{11} &= \sum_{k=1}^5 \frac{E_k}{1-\nu_k^2} \left(h_k z_k^2 + \frac{h_k^3}{12} \right), D_{12} = \sum_{k=1}^5 \frac{\nu_k E_k h_k z_k}{1-\nu_k^2}, D_{22} = D_{11}, \\
 D_{16} &= D_{26} = 0, D_{66} = \sum_{k=1}^n \frac{E_k}{2(1+\nu_k)} \left(h_k z_k^2 + \frac{h_k^3}{12} \right), \\
 \Delta &= A_{22}A_{11} - A_{12}^2, A_{31}^* = \frac{1}{A_{66}}, A_{32}^* = -\frac{B_{66}}{A_{66}}, \\
 A_{11}^* &= \frac{A_{22}}{\Delta}, A_{12}^* = -\frac{A_{12}}{\Delta}, A_{13}^* = \frac{A_{12}B_{12} - A_{22}B_{11}}{\Delta}, A_{14}^* = \frac{A_{12}B_{22} - A_{22}B_{12}}{\Delta}, \\
 A_{22}^* &= \frac{A_{11}}{\Delta}, A_{23}^* = \frac{A_{12}B_{11} - A_{11}B_{12}}{\Delta}, A_{24}^* = \frac{A_{12}B_{12} - A_{11}B_{22}}{\Delta}, \\
 T_{11} &= (B_{11}A_{12}^* + B_{12}A_{22}^*), T_{12} = (B_{12}A_{11}^* + B_{22}A_{12}^*), \\
 T_{13} &= [(B_{11}A_{11}^* + B_{12}A_{12}^*) + (B_{12}A_{12}^* + B_{22}A_{22}^*) - 2B_{66}A_{31}^*], \\
 T_{14} &= +(-B_{11}A_{13}^* - B_{12}A_{23}^* - D_{11}), T_{15} = +(-B_{12}A_{14}^* - B_{22}A_{24}^* - D_{22}), \\
 T_{16} &= + \left[\begin{array}{l} -B_{11}A_{14}^* - B_{12}A_{24}^* - D_{12} - B_{12}A_{13}^* \\ -B_{22}A_{23}^* - D_{12} - 4B_{66}A_{32}^* - 4D_{66} \end{array} \right], \\
 S_1 &= \left(\frac{T_{11}}{A_{22}^*} + \frac{T_{12}}{A_{11}^*} \right) \left(\frac{-8\alpha\beta}{3ab} \right) (W + 2W_0)W, \\
 S_2 &= + \left[(T_{11}\alpha^4 + T_{12}\beta^4 + T_{13}\alpha^2\beta^2) \frac{[A_{14}^*\beta_n^4 + A_{23}^*\alpha_m^4 - (2A_{32}^* - A_{13}^* - A_{24}^*)\alpha_m^2\beta_n^2]}{[A_{11}^*\beta_n^4 + A_{22}^*\alpha_m^4 + (A_{12}^* + A_{12}^* + A_{31}^*)\alpha_m^2\beta_n^2]} \right] W, \\
 S_3 &= -\frac{\beta^4}{8A_{22}^*} (W^2 + 2WW_0)(W + W_0), \\
 S_4 &= \beta\alpha \frac{4}{ab} \left(\frac{32}{9} - \frac{8}{9} \right) \frac{A_{14}^*\beta^4 + A_{23}^*\alpha^4 - (2A_{32}^* - A_{13}^* - A_{24}^*)\alpha^2\beta^2}{A_{11}^*\beta^4 + A_{22}^*\alpha^4 + (A_{12}^* + A_{12}^* + A_{31}^*)\alpha^2\beta^2} (W + W_0)W, \\
 S_5 &= (T_{14}\alpha^4 + T_{15}\beta^4 + T_{16}\alpha^2\beta^2)(W + W_0), S_6 = \frac{4}{ab}.
 \end{aligned}$$

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